


# **APOLLO X-BEAM RIDGE PIECES DESIGN CALCULATIONS**

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CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			 ALAN WHITE DESIGN
	Element : Brief			
	Job Number : G0041	By : anw	Date: Apr 03	
	Document No : 003B	Checked :	Date:	

**Brief**

The brief is to prepare calculated values for the capacity of the Apollo X-BEAM ridge connection to BS 8118.

The beams are fabricated from tube extrusions in aluminium alloy 6082 T6

**Design resistance**

The design resistance of the apollo lattice beam is

**Based on BS8118**

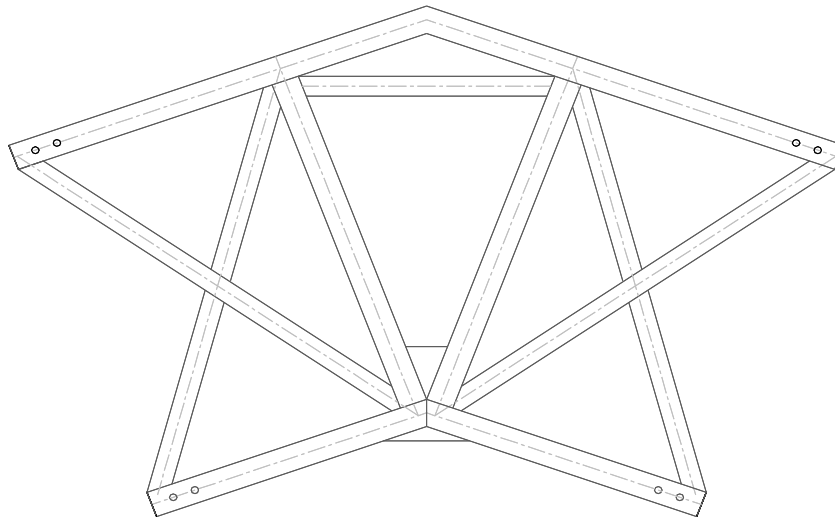
Max moment on the beam is

<b>Allowable moment</b>	<b>42.9 kNm</b>
<b>Ultimate moment</b>	<b>57.0 kNm</b>

and Maximum Shear is

<b>Allowable shear</b>	<b>45.4 kN</b>
<b>Ultimate shear</b>	<b>60.4 kN</b>

**Geometry**



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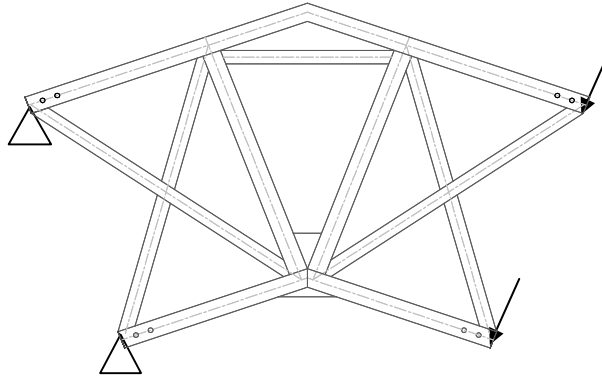
**Loading**

For the shear load case the applied forces are

$$V = 45.4/2$$

$$= 22.7 \text{ kN}$$

These are applied at the top and bottom booms as below

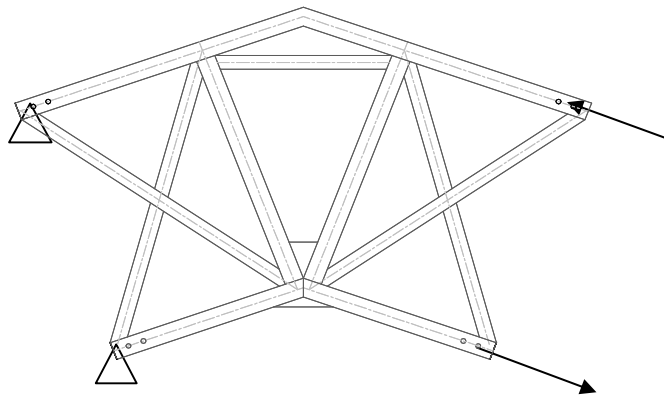


The model has the ridge piece supported as shown on the top and bottom booms by pins


For the moment load case

$$P = 42.9/0.7$$

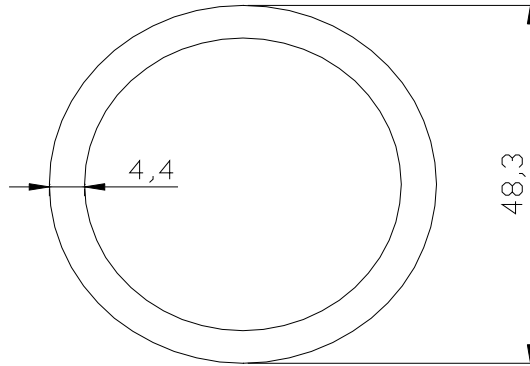
$$= 61.29 \text{ kN}$$



The direction of the forces are interchangeable and can produce either tension or compression in the members.

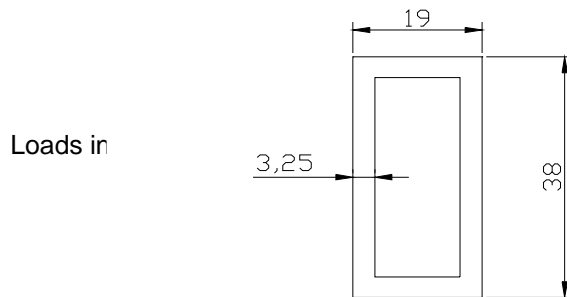
CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			 ALAN WHITE DESIGN
	Element : Section Properties			
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**Main boom and verticals**




Area:	606.83 mm <sup>2</sup>
Bounding box:	X: -24.15 - 24.15 mm Y: -24.15 - 24.15 mm
Moments of inertia:	X: 147654.64 mm <sup>4</sup> Y: 147654.64 mm <sup>4</sup>
Radii of gyration:	X: 15.60 mm Y: 15.60 mm
Elastic Modulus	X: 6114.06 mm <sup>3</sup> Y: 6114.06 mm <sup>3</sup>
Plastic Modulus	X: 8253.99 mm <sup>3</sup> Y: 8253.99 mm <sup>3</sup>

**Diagonals**




Area:	328.25 mm <sup>2</sup>
Bounding box:	X: -9.5 -- 9.5 mm Y: -19.00 -- 19.00 mm
Moments of inertia:	X: 54322.46 mm <sup>4</sup> Y: 16593.21 mm <sup>4</sup>
Radii of gyration:	X: 12.86 mm Y: 7.11 mm
Elastic Modulus	X: 2859.08 mm <sup>3</sup> Y: 1746.65 mm <sup>3</sup>
Plastic Modulus	X: 3758.22 mm <sup>3</sup> Y: 2199.03 mm <sup>3</sup>

**Material**

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The alloy 6082T6 has the following properties

$p_0$ =	255 N/mm <sup>2</sup>
$p_a$ =	280 N/mm <sup>2</sup>
$p_v$ =	155 N/mm <sup>2</sup>

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			
	Element : Main boom			
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**Classification**

4.3.1

$$\begin{aligned} \beta &= 3*((D/t)^{0.5}) \\ &= 3*((48.3-4.4)/4.4)^{0.5} \\ &= 9.48 \end{aligned}$$

$$\begin{aligned} \epsilon &= (250/p_0)^{0.5} \\ &= (250/255)^{0.5} \\ &= 0.99 \end{aligned}$$

$$\begin{aligned} \beta_1 &= 15\epsilon \\ &= 15*0.99 \\ &= 14.85 \\ &> 9.48 \end{aligned}$$

**Section is compact**

**Bending capacity** 4.5.2.2

$$\begin{aligned} M_{rs} &= p_0 S_n / \gamma_m & p_0 &= 255 \text{ N/mm}^2 \\ & & S_n &= 8.25 \text{ cm}^3 \\ & & \gamma_m &= 1.2 \\ &= 255 * 8.25 / 1200 \\ &= \mathbf{1.74 \text{ kNm}} \end{aligned}$$

**Shear** 4.5.3.2

$$\begin{aligned} V_{rs} &= p_v A_v / \gamma_m & p_v &= 155 \text{ N/mm}^2 \\ & & A_v &= 0.6A \\ & & &= 0.6 * 606.83 \\ & & &= 364.1 \text{ mm}^2 \\ & & \gamma_m &= 1.2 \\ &= 155 * 364.1 / 1200 \\ &= \mathbf{47.03 \text{ kN}} \end{aligned}$$

Loads indicated in italics and shaded are limited by shear.

**Lateral Torsional Buckling**

No check required for CHS


**Tension**

4.6  
for General Tension

$$\begin{aligned} P_{rs} &= p_0 A / \gamma_m & p_0 &= 255 \text{ N/mm}^2 \\ & & A &= 606.83 \text{ mm}^2 \\ & & \gamma_m &= 1.3 \\ &= 255 * 606.83 / 1300 \\ &= \mathbf{119.03 \text{ kN}} \end{aligned}$$

For local at splice

$$\begin{aligned} P_{rs} &= p_a A_n / \gamma_m & p_a &= 280 \text{ N/mm}^2 \\ & & A_n &= A - 2dt \\ & & &= 606.83 - 2 * 14 * 4.4 \\ & & &= 483.63 \text{ mm}^2 \\ & & \gamma_m &= 1.3 \\ &= 280 * 483.63 / 1300 \\ &= \mathbf{104.17 \text{ kN}} \end{aligned}$$

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**Compression**

4.7

$$Pr = psA/\gamma m$$

**for 1m bracing**

$$L = 950.00 \text{ m}$$

$$r = 15.6 \text{ mm}$$

$$\lambda = KL/r = 0.7 \cdot 950 / 15.6 = 42.63 \quad K = 0.7$$

Fig 4.10b gives

$$ps = 184.00 \text{ N/mm}^2$$

$$A = 606.83 \text{ mm}^2$$

$$\gamma m = 1.3$$

$$Pr = 184 \cdot 606.83 / 1300$$

$$= \mathbf{85.89 \text{ kN}}$$

**for 0.5m bracing**

$$L = 450.00 \text{ m}$$

$$r = 15.6 \text{ mm}$$

$$\lambda = KL/r = 0.7 \cdot 450 / 15.6 = 20.19 \quad K = 0.7$$


$$ps = 205.00 \text{ N/mm}^2$$

$$A = 606.83 \text{ mm}^2$$

$$\gamma m = 1.3$$

$$Pr = 205 \cdot 606.83 / 1300$$

$$= \mathbf{95.69 \text{ kN}}$$

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			 ALAN WHITE DESIGN
	Element : Diagonal			
	Job Number : G0041	By : anw	Date: Apr 03	
	Document No : 003B	Checked :	Date:	

### Classification

4.3.1

$$\begin{aligned}\beta &= 3*((D/t)^{0.5}) \\ &= 3*((38-3.25)/3.25)^{0.5} \\ &= 9.81\end{aligned}$$

$$\begin{aligned}\epsilon &= (250/p_0)^{0.5} \\ &= (250/255)^{0.5} \\ &= 0.99\end{aligned}$$

$$\begin{aligned}\beta_1 &= 15\epsilon \\ &= 15*0.99 \\ &= 14.85 \\ &> 9.81\end{aligned}$$

Section is compact

### Bending capacity 4.5.2.2

$$\begin{aligned}M_{rs} &= p_0 S_n / \gamma_m & p_0 &= 255 \text{ N/mm}^2 \\ & & S_n &= 3.76 \text{ cm}^3 \\ & & \gamma_m &= 1.3 \\ &= 255 * 3.76 / 1300 \\ &= \mathbf{0.74 \text{ kNm}}\end{aligned}$$

### Shear 4.5.3.2

$$\begin{aligned}V_{rs} &= p_v A_v / \gamma_m & p_v &= 155 \text{ N/mm}^2 \\ & & A_v &= 0.8 N D t \\ & & &= 0.8 * 2 * 38 * 3.25 \\ & & &= 197.6 \text{ mm}^2 \\ & & \gamma_m &= 1.3 \\ &= 155 * 197.6 / 1300 \\ &= \mathbf{23.56 \text{ kN}}\end{aligned}$$

Loads indicated in italics and shaded are limited by shear.

### Lateral Torsional Buckling

$$\text{overall length } L = \frac{\text{SQRT}(652^2 + 425^2)}{778.29}$$

length to brace point is L/2

$$\begin{aligned}L_e &= 0.85 * 778.29 / 2 \\ &= 330.77\end{aligned}$$

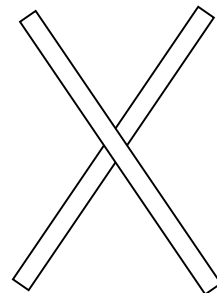
$$\begin{aligned}\lambda &= L_e / r_y \\ &= 330.77 / 7.11 \\ &= 47\end{aligned}$$


$$p_s = 220 \text{ N/mm}^2$$

$$M_{rx} = p_s S / \gamma_m$$

$$\begin{aligned}&= 220 * 3.76 / 1200 \\ &= \mathbf{0.69 \text{ kNm}}\end{aligned}$$

$$\begin{aligned}S &= 3.76 \text{ cm}^3 \\ \gamma_m &= 1.2\end{aligned}$$



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**Tension**

4.6

for General Tension only ( no local holes)

$$\begin{aligned}
Prs &= p_o A / \gamma_m & p_o &= 255 \text{ N/mm}^2 \\
& & A &= 328.3 \text{ mm}^2 \\
& & \gamma_m &= 1.2 \\
& & & = 255 * 328.3 / 1200 \\
& & & = \mathbf{69.76 \text{ kN}}
\end{aligned}$$

**Compression**

4.7

$$\begin{aligned}
Pr &= p_s A / \gamma_m \\
L &= 0.338 \text{ m} \\
r &= 7.11 \text{ mm} \\
\lambda &= KL/r & K &= 0.7 \\
&= 0.7 * 338 / 7.11 \\
&= 33.28
\end{aligned}$$


Fig 4.10b gives

$$\begin{aligned}
ps &= 182.00 \text{ N/mm}^2 & A &= 328.3 \text{ mm}^2 \\
& & \gamma_m &= 1.2 \\
Pr &= 182 * 328.3 / 1200 \\
&= \mathbf{49.79 \text{ kN}}
\end{aligned}$$

for local squashing

$$\begin{aligned}
Prs &= p_a A_e / \gamma_m & p_a &= 280 \text{ N/mm}^2 \\
& & A_e &= 164.2 \text{ mm}^2 \\
& & \gamma_m &= 1.2 \\
& & & = 280 * 164.2 / 1200 \\
& & & = \mathbf{38.31 \text{ kN}}
\end{aligned}$$

Use local squashing value

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**Tension**


4.6

for General Tension only ( no local holes)

$$\begin{aligned}
 \text{Prs} &= p_o A / \gamma_m & p_o &= 255 \text{N/mm}^2 \\
 & & A &= 328.3 \text{mm}^2 \\
 & & \gamma_m &= 1.3 \\
 &= 255 * 328.3 / 1300 \\
 &= \mathbf{64.40 \text{ kN}}
 \end{aligned}$$

for local softening


$$\begin{aligned}
 \text{Prs} &= p_a A_e / \gamma_m & p_a &= 280 \text{N/mm}^2 \\
 & & A_e &= 164.2 \text{mm}^2 \\
 & & \gamma_m &= 1.2 \\
 &= 280 * 164.2 / 1200 \\
 &= \mathbf{38.31 \text{ kN}}
 \end{aligned}$$

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			
	Element : Shear load - 10 degree ridge			
	Job Number : G0041	By : anw	Date: Apr 03	
	Document No : 003B	Checked :	Date:	

Shear load - 10 degree ridge

Restrained at 0.5m


Element	Action	Formula	Ultimate	Calculated	Factor
Boom	Moment	Mrs	1.74	0.24	7.23
	Shear	Vrs	47.03	1.85	25.42
	Axial	Pry	95.69	78.89	1.21
		coexist M		0.08	
	Combined	$P/Prs+M/Mrs < 1$		0.87	1.15
Vertical	Moment	Mrs	1.74	0.17	10.21
	Shear	Vrs	47.03	0.45	104.51
	Axial	Pry	95.69	32.8	2.92
		coexist M		0.17	
	Combined	$P/Prs+M/Mrs < 1$		0.44	2.27
Diagonal	Axial	Prs	38.31	35.2	1.09
				<b>Factor</b>	<b>1.09</b>

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			
	Element : Moment load - 20 degree ridge			
	Job Number : G0041	By : anw	Date:Apr 03	
	Document No : 003B	Checked :	Date:	

Moment load - 20 degree ridge

Restrained at 0.5m


Element	Action	Formula	Ultimate	Calculated	Factor
Boom	Moment	Mrs	1.74	0.13	13.35
	Shear	Vrs	47.03	0.28	167.96
	Axial	Pry	95.69	87.39	1.09
		coexist M		0.04	
	Combined	P/Prs+M/Mrs<1		0.94	1.07
Vertical	Moment	Mrs	1.74	0.03	57.87
	Shear	Vrs	47.03	0.07	671.85
	Axial	Pry	95.69	14.6	6.55
		coexist M		0.03	
	Combined	P/Prs+M/Mrs<1		0.17	5.89
Diagonal	Axial	Prs	38.31	2.42	15.83
				<b>Factor</b>	<b>1.07</b>

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			 ALAN WHITE DESIGN
	Element : Shear load - 20 degree ridge			
	Job Number : G0041	By : anw	Date: Apr 03	
	Document No : 003B	Checked :	Date:	

Shear load - 20 degree ridge

Restrained at 0.5m


Element	Action	Formula	Ultimate	Calculated	Factor	
Boom	Moment	Mrs	1.74	0.28	6.20	
	Shear	Vrs	47.03	2.14	21.98	
	Axial	Pry		95.69	91.99	1.04
		coexist M			0.07	
	Combined	P/Prs+M/Mrs<1			1.00	1.00
Vertical	Moment	Mrs	1.74	0.18	9.65	
	Shear	Vrs	47.03	0.49	95.98	
	Axial	Pry		95.69	29.1	3.29
		coexist M			0.06	
	Combined	P/Prs+M/Mrs<1			0.34	2.95
Diagonal	Axial	Prs	38.31	34.6	1.11	
				<b>Factor</b>	<b>1.00</b>	

CALCULATION SHEET	Project : Apollo X-BEAM -Ridge Piece			 ALAN WHITE DESIGN
	Element : Moment load - 20 degree ridge			
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	Document No : 003B	Checked :	Date:	

Moment load - 20 degree ridge

Restrained at 0.5m

Element	Action	Formula	Ultimate	Calculated	Factor
Boom	Moment	Mrs	1.74	0.51	3.40
	Shear	Vrs	47.03	3.64	12.92
	Axial	Pry	95.69	83.74	1.14
		coexist M		0.19	
	Combined	P/Prs+M/Mrs<1		0.98	1.02
Vertical	Moment	Mrs	1.74	0.05	34.72
	Shear	Vrs	47.03	0.08	587.87
	Axial	Pry	95.69	28.91	3.31
		coexist M		0.11	
	Combined	P/Prs+M/Mrs<1		0.36	2.78
Diagonal	Axial	Prs	38.31	4.9	7.82
				<b>Factor</b>	<b>1.02</b>

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**Sections**

Both the 20 degree and the 10 degree ridge pieces have been checked.

**Results**

The design has been to the code BS8118 under loading that produces equivalent moments and shears to the rated values of the X-beam.

The resistance of the ridge piece is greater than required to The factor below is strength of ridge piece/ strength of X-beam

20 degree

Factor for bending	<b>1.00</b>
Factor for shear	<b>1.03</b>

10 degree

Factor for bending	<b>1.07</b>
Factor for shear	<b>1.09</b>

**Conclusions**

**This geometry for the ridge pieces is equivalent in strength to the X-beams and suitable to form the ridge joints.**